

AeroGraphics

The Original Aircraft Lettering Company



Cessna 205, 206, 207 Interior Kit PAGE 1 of 2

NOTE: Modifications and changes to accommodate your specific aircraft will be made at **NO EXTRA CHARGE**.
Partial kits available upon request.

IF YOU HAVE ANY QUESTIONS PLEASE CALL!

Phone: 1-800-336-9633 www.aerographics.com Fax: 1-970-461-2065

WINTERIZATION KIT MUST BE REMOVED WHEN OUTSIDE AIR TEMPERATURE IS ABOVE 20° F.

BAGGAGE NET 180 LBS. MAX CAPACITY REFER TO WEIGHT AND BALANCE DATA FOR BAGGAGE AND CARGO LOADING

AVOID OPERATION BETWEEN 1850 AND 2150 RPM ABOVE 24 IN. M.P.

EMERGENCY LOCATOR TRANSMITTER INSTALLED AFT OF THIS PARTITION. MUST BE SERVICED IN ACCORDANCE WITH FAR 91.52

CAUTION
LEAKING FUEL CAPS CAN CAUSE LOSS OF FUEL AND ERRONEOUSLY HIGH FUEL QUANTITY INDICATIONS.

CONTROL LOCK REMOVE BEFORE STARTING ENGINE

EMERGENCY EXIT OPERATION
1. OPEN FWD CARGO DOOR AS FAR AS POSSIBLE
2. ROTATE RED LEVER IN REAR CARGO DOOR FWD.
3. FORCE REAR CARGO DOOR FULL OPEN

WITH CARGO DOORS REMOVED DO NOT EXCEED 130 KIAS

MAXIMUM ALTITUDE LOSS DURING ELECTRIC TRIM MALFUNCTION - 200 FT.

AEROGRAPHICS
1405 N. Prospect Dr
Loveland, CO 80538
800-336-9633

CESSNA
205-206-207 U
INTERIOR KIT
1 OF 2

TURN STROBE LIGHTS OFF WHEN TAXIING IN VICINITY OF OTHER AIRCRAFT, OR WHEN FLYING IN FOG OR CLOUDS. STANDARD POSITION LIGHTS TO BE USED FOR ALL NIGHT OPERATIONS

THIS AIRPLANE MAY BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF PLACARDS, MARKINGS, AND MANUALS.

MAXIMUMS

U206 / U206	U206
LANDPLANE	FLOATPLANE
MANEUVERING SPEED (IAS) 120 KTS.	120 KTS.
GROSS WEIGHT 3500 LBS.	3500 LBS.
FLIGHT LOAD FACTOR FLAPS UP +3.8, 1.52	FLAPS DOWN +2

NO AEROBATIC MANEUVERS, INCLUDING SPINS, APPROVED. ALTITUDE LOSS IN STALL RECOVERY 240 FT. FLIGHT INTO KNOWN ICING CONDITIONS PROHIBITED. THIS AIRPLANE IS CERTIFIED FOR THE FOLLOWING FLIGHT OPERATIONS AS OF DATE OF ORIGINAL AIRWORTHINESS CERTIFICATE.
DAY - NIGHT - VFR - IFR

ALTIMETER ENCODER EQUIPPED

MAX PERMISSIBLE AUTOPILOT SPEED DURING AUTOPILOT MALFUNCTIONS RECOVERY: 300 FEET

MANUAL PRIMER

PULL OFF

CO-PILOT

PILOT

IDENT

REPLY

BOOST

CLIMB FUEL FLOW

S.L.	22 GPH
4000 FT.	20 GPH
8000 FT.	18 GPH

CO-PILOT

PILOT

IDENT

REPLY

BOOST

OXYGEN REFILL

PRIMER

INTER-COM

STRAINER DRAIN PULL

NOSE DOWN

ELEVATOR TAKE OFF

TRIM NOSE UP

ON HI ON

EMERG

START

ALT MASTER

AUX FUEL PUMP

WATERIZATION KIT MUST BE REMOVED WHEN OUTSIDE AIR TEMPERATURE IS ABOVE 20° F.

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THIS AIRPLANE MAY BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF PLACARDS, MARKINGS, AND MANUALS.

MAXIMUMS

U206 / U206	U206
LANDPLANE	FLOATPLANE
MANEUVERING SPEED (IAS) 120 KTS.	120 KTS.
GROSS WEIGHT 3500 LBS.	3500 LBS.
FLIGHT LOAD FACTOR FLAPS UP +3.8, 1.52	FLAPS DOWN +2

NO AEROBATIC MANEUVERS, INCLUDING SPINS, APPROVED. ALTITUDE LOSS IN STALL RECOVERY 240 FT. FLIGHT INTO KNOWN ICING CONDITIONS PROHIBITED. THIS AIRPLANE IS CERTIFIED FOR THE FOLLOWING FLIGHT OPERATIONS AS OF DATE OF ORIGINAL AIRWORTHINESS CERTIFICATE.
DAY - NIGHT - VFR - IFR

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BAGGAGE NET 180 LBS. MAX CAPACITY REFER TO WEIGHT AND BALANCE DATA FOR BAGGAGE AND CARGO LOADING

AVOID OPERATION BETWEEN 1850 AND 2150 RPM ABOVE 24 IN. M.P.

EMERGENCY LOCATOR TRANSMITTER INSTALLED AFT OF THIS PARTITION. MUST BE SERVICED IN ACCORDANCE WITH FAR 91.52

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3. FORCE REAR CARGO DOOR FULL OPEN

WITH CARGO DOORS REMOVED DO NOT EXCEED 130 KIAS

MAXIMUM ALTITUDE LOSS DURING ELECTRIC TRIM MALFUNCTION - 200 FT.

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REFER TO WEIGHT & BALANCE DATA FOR BAGGAGE / CARGO LOADING NEVER EXCEED 300 LBS CARGO WEIGHT. COWL FLAP EXTENSIONS MUST BE INSTALLED WITH CARGO PACK.

AIR TEMP COOLER ON OFF MAX- +

AIR CONDITIONING FAN HI MED LOW OFF

NO SMOKING

DO NOT TURN OFF ALTERNATOR IN FLIGHT EXCEPT IN EMERGENCY

AEROGRAFICS
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CESSNA
205-206-207 U&T
INTERIOR KIT
2 OF 2

THIS AIRPLANE MUST BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS AS STATED IN THE FORM OF PLACARDS, MARKINGS AND MANUALS.
NO AEROBATIC MANEUVERS, INCLUDING SPINS, APPROVED.

MAXIMUMS
MANEUVERING SPEED 148 MPH - CAS ALTITUDE LOSS IN STALL RECOVERY 350 FT
DESIGN WEIGHT 3800 LBS
FLIGHT MANEUVERING LOAD FACTORS FLAPS UP +3.8, -1.52 FLAPS DOWN +2.4
FLAP EXTENSION SPEEDS 110 MPH CAS 0°-30° 160 MPH CAS 0°-10°
THIS AIRPLANE IS CONTROLLABLE IN 20 KNOTS CROSS WIND. KNOWN ICING CONDITIONS TO BE AVOIDED. THIS AIRPLANE IS CERTIFIED FOR THE FOLLOWING FLIGHT OPERATIONS AS OF DATE OF ORIGINAL AIRWORTHINESS CERTIFICATE.
VFR - IFR - DAY - NIGHT

TAKEOFF CHECKLIST

FUEL ON PROPER TANK ENGINE GAGES CHECKED DOORS LATCHED SEAT LOCKED	PRIMER LOCKED MIXTURE RICH PROPELLER SET FASTEN BELTS/HARNESSES	FLAPS SET TRIM TAB SET CONTROLS FREE AIR COND OFF
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LANDING CHECKLIST

SEAT BACKS ERECT FASTEN BELTS/HARNESSES FUEL ON PROPER TANK	SEAT LOCKED MIXTURE RICH	PROPELLER SET FLAPS SET AIR CONDITIONER OFF
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ATTENDANT MUST FACE FORWARD DURING TAKE OFF & LANDING

DO NOT OPEN WINDOW ABOVE 105 KTS OR WHEN USING ALTERNATE STATIC SOURCE

FASTEN SEAT BELTS **FASTEN SEAT BELTS** **POST - BOTH - FLOOD PANEL LIGHTS** **WING PULL ON**

WITH CARGO DOORS REMOVED DO NOT EXCEED 170 MPH IAS **WITH CARGO DOORS REMOVED DO NOT EXCEED 170 MPH IAS** **WITH CARGO DOORS REMOVED DO NOT EXCEED 150 MPH IAS** **WITH CARGO DOORS REMOVED DO NOT EXCEED 170 MPH IAS**

USE "HI" FOR EMERG ONLY **TIRE PRESSURE** MAIN (800X6) 35 PSI NOSE (800X5) 28 PSI **MANEUVERING SPEED 120 IAS** **AREA NAVIGATION USE LIMITED TO VFR ONLY** **OXYGEN PULL ON** **WING UP 140°** **VAC. LIGHTS**

MAX PERMISSIBLE AUTOPILOT SPEED 180 MPH CAS. MAX ALTITUDE LOSS DURING AUTOPILOT MALFUNCTION RECOVERY 200 FEET **ATTENDANT MUST FACE FORWARD DURING TAKEOFF AND LANDING** **WING FLAPS 10°** **PUSH TEST**

MAXIMUM ALTITUDE LOSS DURING ELECTRIC TRIM MALFUNCTION - 330 FT. **TIRE PRESSURE** MAIN (600X6) 55 PSI NOSE (500X5) 49 PSI **MAXIMUM ALTITUDE LOSS DURING ELECTRIC TRIM MALFUNCTION - 330 FT.** **TIRE PRESSURE** MAIN (600X6) 42 PSI NOSE (500X5) 45 PSI **ATTENDANT MUST FACE FORWARD DURING TAKEOFF AND LANDING** **WING FLAPS 20°** **ROT BEACON**

BAGGAGE NET 120 LBS. MAX CAPACITY REFER TO WEIGHT AND BALANCE DATA FOR BAGGAGE AND CARGO LOADING **WING FLAPS FULL** **ROT BEACON**

NO BAGGAGE THIS AREA

MAJOR FUEL FLOW FLUCTUATIONS / POWER SURGES
1. ADJ. FUEL PUMP ON, ADJUST MIXTURE
2. SELECT OPPOSITE TANK
3. WHEN FUEL FLOW STABILIZES RESUME NORMAL OPERATIONS
SEE P.O.H. FOR EXPANDED INSTRUCTIONS

THIS AIRPLANE MUST BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS AS STATED IN THE FORM OF PLACARDS, MARKINGS AND MANUALS.
NO AEROBATIC MANEUVERS, INCLUDING SPINS APPROVED.

MAXIMUMS
MANEUVERING SPEED 148 M.P.H. (CAS)
GROSS WEIGHT 3800 LB.
FLIGHT MANEUVERING LOAD FACTORS:
FLAPS UP +3.8; -1.52 FLAPS DOWN +2.4
ALTITUDE LOSS IN STALL RECOVERY 350 FT.
FLAP EXTENSION SPEED 110 M.P.H. (CAS) 0° - 30°
160 M.P.H. (CAS) 0° - 10°

AIRCRAFT IS CONTROLLABLE IN 20 KNOT CROSS-WINDS.
KNOWN ICING CONDITIONS TO BE AVOIDED
THIS AIRPLANE IS CERTIFIED FOR THE FOLLOWING FLIGHT OPERATIONS AS OF DATE OF ORIGINAL AIRWORTHINESS CERTIFICATION:
VFR - IFR - DAY - NIGHT

EMERGENCY EXIT OPERATION

1. ROTATE FORWARD CARGO DOOR HANDLE FULL FORWARD HANDLE THEN FULL AFT
2. OPEN FORWARD CARGO DOOR AS FAR AS POSSIBLE
3. ROTATE RED LEVER IN REAR CARGO DOOR FORWARD
4. FORCE REAR CARGO DOOR FULL OPEN

MAX POWER SETTINGS AND FUEL FLOW

ALTITUDE (FT)	1800 RPM	2000 RPM	2200 RPM	2400 RPM	2600 RPM	2800 RPM	3000 RPM
0	55	65	75	85	95	105	115
1000	55	65	75	85	95	105	115
2000	55	65	75	85	95	105	115
3000	55	65	75	85	95	105	115
4000	55	65	75	85	95	105	115
5000	55	65	75	85	95	105	115
6000	55	65	75	85	95	105	115
7000	55	65	75	85	95	105	115
8000	55	65	75	85	95	105	115
9000	55	65	75	85	95	105	115
10000	55	65	75	85	95	105	115

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